

Layered Structure of CoCrAlY/ZrSiO₄/Al₂O₃ Coatings Before and After Thermal Treatment

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In work, dense CoCrAlY/ZrSiO₄/Al₂O₃ coatings have been prepared on nickel based superalloy substrates by a new multi-chamber gas-dynamic accelerator. The coatings were subjected to thermal treatment in vacuum at a temperature of 950 °C for 2 hours. Scanning microscope with an EDX microprobe was used to analyze the chemical composition. Analysis of structural changes, mainly from the aspect of porosity, was also carried out using an optical microscope. Measurement of the microhardness of coatings was done with a microhardness tester DM-8B using a Vicker's hardness tester at a test load of 50 g. A positive impact of thermal treatment on the coating microstructure and microhardness was observed. The CoCrAlY/ZrSiO₄/Al₂O₃ coatings after thermal treatment were observed to be continuous, without cracks, and well bonded with the substrate. The absence of discontinuities in the interfaces among the different layers was indicative of good adhesion between them. Thermal treatment resulted in an increase in the coating hardness.

Keywords: Microstructure, Microhardness, Multi-chamber gas-dynamic accelerator, Nickel alloys, Thermal barrier coatings.

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1. INTRODUCTION

TBCs are finding increasing application in engines which operate with low quality fuels containing vanadium and sulfur. During service at a temperature of 600-1000 °C, vanadate salt and sulfate condense on the surface of TBCs [1]. TBCs have a two layered structure consisting of a bond coat and yttria stabilized zirconia (YSZ) topcoat [2]. Zirconia itself shows good resistance to the attack of the molten sulfate or vanadate compounds arising, yttria is leached out of the zirconia causing the structural destabilization of ZrO₂ (i.e., transformation of the zirconia from the tetragonal and/or cubic to monoclinic phase). The structural destabilization of ZrO₂ is accompanied by a large destructive volume change, leading to large stresses within the YSZ, which eventually results in the delamination and spalling of the coating [3, 4]. Many kinds of ceramic were tested as alternative thermal barrier coating material of the thermal barrier coating system in decades. The zircon (ZrSiO₄) is also chosen as the suitable ceramic material for the thermal barrier coatings [5-7]. It shows excellent thermal shock resistance as a result of its very low thermal expansion coefficient [1]. Over the years attempts have been made to seal the surface of zirconia TBCs using ion and laser beams [8] or various "seal coats" [9] to prevent the penetration of molten deposits into the porous YSZ coating [10]. Alumina has a high melting point and stability without showing phase transition at high temperatures like the ZrO₂ ceramics. Al₂O₃ has a small solubility particularly in molten salt and is expected to show an excellent corrosion resistance. It seems that the use of fine alumina particles as a dense layer over the YSZ and zircon-based coatings can decrease oxygen

diffusivity and salt penetration through this layer [10].

In our study, both the dense zircon-based ceramic coating [11] as TBC and alumina as overlay coating were obtained by a new multi-chamber gas-dynamic accelerator (MCDS) [11, 12]. The effect of thermal treatment (in vacuum) on the microstructure and microhardness of CoCrAlY/ZrSiO₄/Al₂O₃ coatings was investigated using optical microscopy (OM), scanning electron microscopy (SEM), and Vickers hardness tester.

2. EXPERIMENTAL PROCEDURE

Nickel based superalloy (Ni-20Cr-11.5Co-0.6Mo-3W-0.5Ni-3.5Ti-21Al-0.5Fe, all in wt. %) (30 × 30 × 5 mm³) which grit blasted with alumina particles was used as substrate. Three types of powders were selected (see Fig. 1): Co-25Cr-11Al-1Y (CoCrAlY) as bond coat, zircon sand (ZrSiO₄) (naturally available) powder and AMPERIT 740.0 Al₂O₃ powder as TBC. Table 1 indicates the characteristic of the powders, and the parameters of coating spraying are listed in Table 2. In the present study, a vertically mounted multi-chamber and a gas-dynamic accelerator (MCDS) were employed to deposit the coatings on nickel based superalloy substrates. The coatings were deposited with a frequency of 20 Hz. MCDS was equipped with several combustion chambers serially connected by channels. These channels were intended for the expiration of the combustion products. They were converged in the central chamber – nozzle. Filling of all chambers by the combustible mixture was carried out simultaneously. MCDS was equipped with a standard powder feeder (by Metco). Continuous gas-powder jet was supplied along the conduit (4-8 m) divided into portions and inserted into the nozzle [11].

To determine the microstructure and elemental composition of the powders and coatings, scanning electronic

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microscope (SEM) (Quanta 200 3D, Nova NanoSEM 450) and Optical Microscope Olympus GX 51 were performed. Investigations were carried out on polished cross-sections that were normal to the surface. Specimens were transversally cut by spark erosion, mechanically polished and prepared by standard metallographic sample preparation: sectioning, mounting and polishing methods. Porosity was determined by the metallographic method with elements of the qualitative and quantitative analysis of the geometry of the pores using an optical inverted Olympus GX51 microscope [13]. Ten arbitrarily selected micrographs for each experimental point were registered with an optic microscope (bright field, magnified 1000×) using the “SIAMS Photolab” program. Phase composition of the powders was determined by the X-ray phase analysis method (diffractometer Rigaku Ultima IV). Crystalline phases were identified by the ICDD PDF-2 (2008) powder diffraction database. Microhardness of the coatings was determined by an automatic microhardness tester DM-8B (Affri) by Vickers’s test at a test load of 50 g. Indentation was carried out on the

cross-sections of the samples of the coatings. On average, 10 tests were used as an indicator of the coating hardness. A high-temperature vacuum furnace (VHT 8/22-GR, Nabertherm GmbH) was used for heat treatment of coatings in vacuum. The samples were placed in the furnace, air was evacuated, and the furnace was heated at 8 °C/min from 23 to 950 °C. The coatings were cooled to 50 °C at a rate of 7.5 °C/min for 2 h, and then allowed to cool to room temperature outside the furnace.

3. RESULTS AND DISCUSSION

In some cases, heat treatment of thermally sprayed deposits can release residual stresses, decrease their porosity, and improve their microstructure and properties [14]. Heat treatment of thermally sprayed coatings can effectively improve their corrosion resistance by changing their microstructure from lamellar one to a bulk-like, by removing such defects as pores and lamellae boundaries, which are known to impair their performance [15, 16].

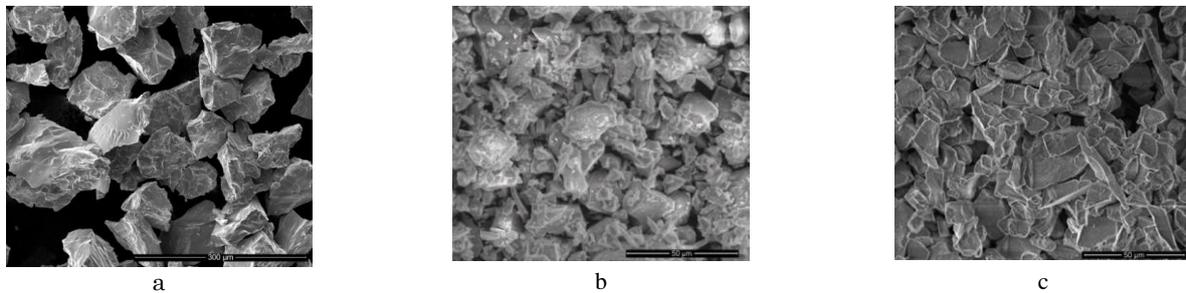


Fig. 1 – Morphology of the CoCrAlY (a), ZrSiO₄ (b), and Al₂O₃ (c) powders (SEM)

Table 1 – The chemical, phase composition, and particle size of the powders

Chemical composition, all in wt. %	Powder			
	Element	CoCrAlY	ZrSiO ₄	Al ₂ O ₃
	Co	59.88	–	–
	Cr	23.24	–	–
	Si	2.66	8.23	0.12
	Fe	1.13	0.46	1.10
	Al	13.10	0.07	54.94
	Zr	–	77.48	–
	O	–	13.77	35.84
C	–	–	8.00	
Particle size distribution, μm				
<i>d</i> (0.1)	6.6	2.6	7.6	
<i>d</i> (0.5)	62.7	18.3	15.8	
<i>d</i> (0.9)	123.4	46.7	29.4	
Identified major phases	Al _{0.94} Co _{1.06} , Cr	Z-zircon (ZrSiO ₄)	γ-Al ₂ O ₃ , α-Al ₂ O ₃ , Fe ₂ O ₃ , SiO ₂	

Table 2 – Spraying parameters of coating layers

Spraying parameters	Coating layer		
	CoCrAlY	ZrSiO ₄	Al ₂ O ₃
Spray distance, mm	55		
Barrel length, mm	500		
Barrel diameter, mm	18	16	16
Powder feed rate, g/h	1400	550	700
Flow rate of fuel mixture components, m ³ /h			
Oxygen	*2.9/**2.6	*4.3/**3.2	*4.3/**3.2

Propane (100 %)	*0.7/**0.7	*0.8/**0.7	*0.8/**0.7
Air	*1.6/**1.1	*0.1/**0.3	*0.1/**0.3
Oxygen/fuel ratio	4.4	5.3	5.3
*Cylindrical form combustion chamber. **Combustion chamber in the form of a disk			

Fig. 2 presents SEM micrographs (back-scattered electron mode) of the polished cross-sections of the CoCrAlY/ZrSiO₄/Al₂O₃ coatings both after deposition and after heat treatment. The dense coatings were formed through the intensive plastic deformation resulting from the impact of a particle moving with high velocity and the tamping effect on the source side of the following particles. The coating thickness is approximately 140 micron; coatings are dense and show a good adherence to the substrate. The thickness of coating layers varied in the range of 25 to 65 microns (Table 3). No major cracking was observed in the microstructures of these coatings both after deposition and after heat treatment. As shown in Fig. 2, the sample had a wavy and stratified

structure with piled-up flattened particles. The lines of laminations are often quite diffuse and poorly defined. The contrast in Fig. 2 indicates that each layer has different chemical compositions. The «coating – substrate» interface and interface between coating layers had no visible macro defects both after deposition and after heat treatment. The metallographic porosity levels in the as-sprayed and heat-treated coatings were very low (see Table 3).

Fig. 2b presents SEM images showing the microstructure of heat-treated coatings. It was found that during high-temperature annealing, the porosity of the CoCrAlY/ZrSiO₄/Al₂O₃ coatings did not change much (see Table 3).

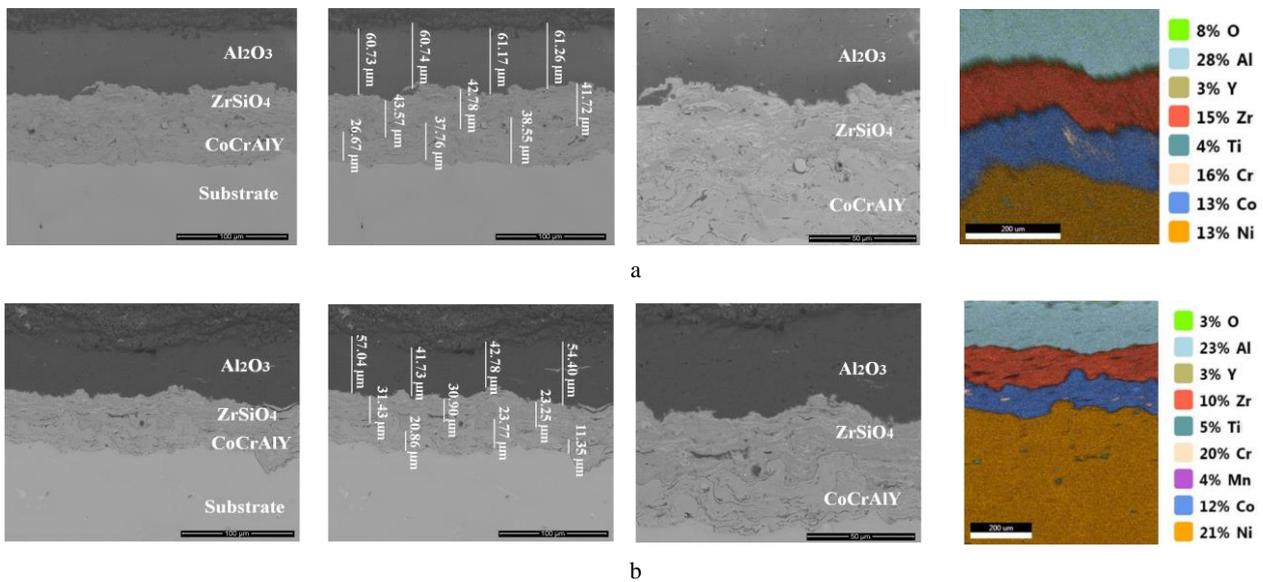


Fig. 2 – SEM micrographs of cross-sections of the CoCrAlY/ZrSiO₄/Al₂O₃ coatings (back-scattered electron mode, Quanta 200 3D) before (a) and after (b) thermal treatment and SEM EDX element distribution maps (NanoSEM 450)

Table 3 – The thickness, porosity, and microhardness of coating layers before and after thermal treatment

Coating layers	Thickness, micron		Porosity, %		Microhardness, HV _{0.05}	
	Thermal treatment					
	Before	After	Before	After	Before	After
Al ₂ O ₃	55 ÷ 65	38 ÷ 60	0.7	0.5	1419 ± 197	1447 ± 253
ZrSiO ₄	30 ÷ 45	15 ÷ 40	0.5	0.3	693 ± 124	829 ± 161
CoCrAlY	25 ÷ 40	11 ÷ 25	0.2	0.1	686 ± 91	788 ± 122

Heat treatment gives rise to no changes in the area of the coating that adjoins the substrate. The absence of cracks or other discontinuities in the interfaces among the different layers of heat treated coatings was indicative of good adhesion between them. The decrease in thickness of coatings (Table 3) owing to heat treatment was due to the active dislocation sources and decrease in the residual stress. The increased compactness with the heat treatment was also predicted to be main reason for good appearance and high hardness [17]. The results showed that the thermal treatment of the coatings led to an increase in the microhardness of CoCrAlY/ZrSiO₄/Al₂O₃ coatings.

4. CONCLUSIONS

It was experimentally demonstrated that the CoCrAlY/ZrSiO₄/Al₂O₃ coatings could be manufactured on nickel based superalloy substrates using a new multi-chamber gas-dynamic accelerator with subsequent thermal treatment of the deposits. The main results can be summarized as follows:

- 1) The coatings presented dense microstructure and porosity below 1.0 % both after deposition and after heat treatment.
- 2) The «coating – substrate» interface and interface between coating layers had no visible macro defects

both after deposition and after heat treatment.

3) The thermal treatment of the coatings leads to an increase of microhardness.

In our next study, both the zircon-based ceramic coating and the $ZrSiO_4/Al_2O_3$ system will be exposed to the molten salt ($Na_2SO_4 + V_2O_5$) at 950 °C. The role of Al_2O_3 in the hot corrosion environment will be examined in greater detail in a forthcoming publication.

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Структура покриття $CoCrAlY/ZrSiO_4/Al_2O_3$ до та після термічної обробки

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У роботі щільні покриття $CoCrAlY/ZrSiO_4/Al_2O_3$ були підготовлені на підкладках із суперсплавів на основі нікелю за допомогою нового багатокамерного газодинамічного прискорювача. Покриття піддавали термічній обробці у вакуумі при температурі 950 °C протягом 2 годин. Для аналізу хімічного складу використовували скануючий мікроскоп з мікронзондом EDX. Аналіз структурних змін, в основному з точки зору пористості, проводили за допомогою оптичного мікроскопа. Вимірювання мікротвердості покриття проводили за допомогою мікротвердоміра DM-8B з використанням пірамідки Вікерса при навантаженні 50 г. Встановлено, що термічна обробка позитивно впливає на мікроструктуру і мікротвердість покриття. Покриття $CoCrAlY/ZrSiO_4/Al_2O_3$ після термічної обробки суцільні, без тріщин і добре пов'язані з підкладкою. Відсутність пор на межах розділу між різними шарами свідчить про хорошу адгезію між ними. Термічна обробка призвела до підвищення твердості покриття.

Ключові слова: Мікроструктура, Мікротвердість, Багатокамерний газодинамічний прискорювач, Нікелеві сплави, Термобар'єрні покриття.